

March 15, 1930

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# AVIATION

*The Oldest American Aeronautical Magazine*



REQUIREMENTS OF *Intermediate* LANDING FIELDS

THE JOB OF *Selling* AIRPLANES

REGULATING AIR COMMERCE—THE *Medical* SECTION

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THE STANDARD FOR AIRCRAFT FINISHES SINCE 1913

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The Oldest American Aeronautical Magazine

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SIX WEEK

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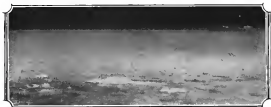
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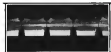


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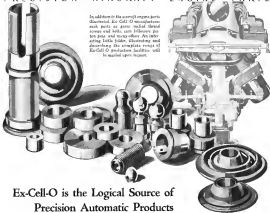
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# AVIATION

THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

A McGraw-Hill Publication ESTABLISHED 1911

EDWARD P. WARNER, Editor

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## Glider Regulation

**E**NTHUSIASM for the glider gives ground support. It stands a respected and a welcome member of the aeronautical family. It is, in fact, in a little danger of their spilling over-enthusiasm that traditionally measures the most lately arrived member of a human family. Moreover flight has taken such a hold upon the imaginations of airplane pilots who have tried it that some of them seem almost to have forgotten their interest in the engine-propelled machine. Thanks to its natural attractions; thanks to aeronautical performances in Germany; thanks to the work of the National Glider Association and other concerned and enthusiastic promoters, thanks especially to the active participation of Colonel and Mrs. Lindbergh, Miss Eberhart, Frank Hawks and other noted pilots, the glider has caught the public fancy. News of gliding aeronautism is rapidly received and eagerly consumed. We are reaching the point where an excess of unguided enthusiasm may become a menace. There is danger lest uncontrolled gliding experiments by a band of unsanctioned individuals may lead to results as disastrous as a new fire-breeding will rise up as almost a result. Unless checked it will rise in the right place and in the right manner, with all possible moderation but with determination and geometry the situation may get so out of hand as to call for stringent measures of repression.

Regulation of the glider and its operator is urgently needed in the interest of safety. Good! How much regulation? What kind of regulation? How applied?

To these questions there is no complete answer. Even if it were agreed (which it is not, and we should be the first to dissent from such a proposal) that the regulation of gliders should follow a line strictly parallel to that imposed upon the airplane, with factory inspection supplied and approval type certificates granted after the same fashion, the emergency is not at hand for writing unworkable regulations.

To derive even the simplest rules for the structural design of gliders and for the load factors to be used, there should be in the first instance a very careful study of the records of the observations made at Doctor George's remarkable institute for soaring flight research in the Rhine Mountains; followed by the prosecution of flight tests and systematic studies in this country with the co-operation of the National Advisory Committee for Aeronautics. So far as we know, no glider has ever carried a modern accelerometer to determine the load that its structure sustained in flight. That should be tried, and tried promptly, both in hard training glides and in prolonged soaring flights such as have been made by Mr. Boush.

Without the need of waiting for specialized research, however, some things are clearly apparent. Gliders are being built by a considerable number of amateur enthusiasts. Some of them are proceeding in complete and horrifying ignorance of the first principles of aircraft construction and of the precautions that it demands. We have seen some of them. We have seen machines with landing wires that flapped, and with tail surfaces that wobbled crazily as the breeze. It needs no knowledge of flight test data or structural loads nor any acquaintance with the theories of stress analysis, to lead to the decision that such machines ought to be obliterated before they obliterate themselves and their pilots. At least so far, the Department of Commerce can go at once. Almost all the important glider clubs are forced now to present a written order, where the Department's inspectors are either regular residents or frequent visitors. Glider licensing might well be undertaken, at least as a stop-gap to meet an emergency, upon the basis of a more superficial external inspection by the Department's representatives to determine whether or not the general appearance and finish justify the presumption of competent workmanship. Such an inspection in any one

of the Aeronomics Branch's staff could give in any instance should be a prerequisite for flying. It is to be noted by experimenters that the glider shall either have been built from blueprints secured from some competent organization or, if from original designs, that the designer shall give evidence at least of knowing what a stress analysis is and of having made some pretense of preparing one, by far the greater part of the seasonal business will be cancelled. Suppliment that with restrictions on amateur participation in such abominably dangerous pursuits as airplane-ventured flight, and we shall be able to breathe more easily about the glider's future.

To most people, at least in most of these outside the airplane industry, the regulation of gliders must appear a small problem and a relatively simple one. We live close enough to it to recognize that that is far from the fact.

In present some novel and perplexing dilemmas to the Department of Commerce, which faces the twin hazards of undue laxity of control and a wave of fatal accidents on the one hand, of undue severity and a stifling of glider development and especially of the initiative of the amateur builder on the other. The second danger is only a little less acute than the first. The glider offers an opportunity of getting the amateur sport back into aeronautical engineering. Amateur invention worked wonders for radio. It can do much for aeronomics. The Department of Commerce has to control the sheep apart from the goats, distinguishing between the amateur who knows what he is about and deserves encouragement, and the amateur whose ignorance or indifference make his activities nothing more than a spectacle and expensive form of self-destruction. We hope to see Secretary Young and his associates equally vigilant against the goats that are in an either side of their true nature. They can depend upon the support of the aeronautical industry for every reasonable measure of control.

//

### Sir Hugh Trenchard

IN August 1912 a course in the Royal Aero Club, returned to fly at the Bristol School at Brooklands, England, and upon graduation was awarded Royal Aero Club Certificate No. 270. On Dec. 31, 1929, as Member of the Royal Air Force he named the post of Chief of the Air Staff. During that period of seventeen years that span, Sir Hugh Trenchard, Bart, GCB, DSO, did more for the development of British military aeroplanes than any other man in the Empire.

When war broke out in August 1914 it was Sir Hugh Trenchard who was given the task of building up at home a Flying Corps for service in the various fronts. Before the year was out he himself was summoned to France to take command of No. 1 Wing of the Royal

Flying Corps. In August 1915 he was appointed to command the Royal Flying Corps on active service, and from that time until December 1917 the success of the R.F.C. in France was his success. At the beginning of 1916 he was recalled from the front and appointed a member of the Air Council, and the first Chief of the Air Staff. His ideas on how to conduct the service and those of Lord Rothermere's, England's first Secretary for State for Air, did not coincide. As a result Sir Hugh Trenchard resigned his post. Lord Rothermere's resignation followed shortly thereafter.

Almost immediately Sir Hugh Trenchard was asked to accept his former post. He also offered the command of the then Royal Air Force in France. Being a man of sterling character he refused to seek any man who had succeeded him to positions of high rank.

The result was that after consultation with Lord Rothermere's successor, Sir Hugh Trenchard consented to take command of what was later termed the Independent Air Force, a group of bombing place squadrons stationed in the east of France, so near the lines as possible and working independently of the troops as the ground. He remained in that command until the Armistice, and the work of the units under him is Royal Air Force history that will never be forgotten.

The beginning of 1919 saw him once again the Chief of the Air Staff, and with perhaps an even greater task to accomplish. At the end of the War England had an air service second to none, but left everything else military it was torn apart while peace came to the land. To Sir Hugh Trenchard again fell the task of building up England's third arm of defense, but this time under peace conditions and the financial opposition of various factions. The Royal Air Force of today, regarded as being capable of protecting not only the British Isles but the entire British Empire from invasion by any other air force, is the result of eleven years of Trenchard leadership.

To write of the hundreds of incidents that brought out the splendid qualities of the man would be to fill all the pages of this issue. To use two expressions quite common at the time of the War, Sir Hugh Trenchard was a "pukka" officer, and a "brass hat" that everyone respected and admired from the least-in-command to the lowest machine. His always had a word of encouragement for the junior officers. He never failed to give credit where credit was due. He had a less sense of humor that could be appreciated by others. He never gave a command that he would not be willing to execute himself. He did his best for his king and country, and he expected the same high degree of effort and loyalty of those under him. We got it.

And now he has returned to "civilian" and a well earned rest from active duty. Just how inactive he will be is problematical. . . . the Royal Air Force will always be appreciative of his thoughts.

However, we take this opportunity of offering our congratulations to a great citizen for a great service

rendered. His record as Sir Hugh Trenchard, "after commencing," will always stand out in British aeronautical history, and his record as Sir Hugh Trenchard, the man, will always be remembered by those who had the good fortune to serve under him.

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### Government Specifications

THE FEDERAL government system of purchasing parts or materials by competitive bidding under established specifications renders a service to aviation that is seldom understood or appreciated by the industry at large. Too often the specification system is condemned by one unfortunate "breakdown," usually more intimately associated with the necessarily confidential nature of Government purchase as a whole or with the inspection personnel than with the specification system itself.

Specifications, properly drawn, provide the basic foundations on which all true progress is reared. A good specification summarizes basic requirements, and the tests necessary to establish compliance with these established requirements.

It encourages fair competition, in that products must meet the minimum requirements before they can compete on a basis of cost. Beyond that, it gives the consumer nearly insurmountable a given field a worthy objective toward which to strive. For the organization that is endeavoring to better their product, it gives them a scale by which to measure their improvement, and if appreciable improvement is made there is an implied promise to revise the specification to meet the new standard of excellence or performance. To the busy commercial airplane manufacturer, it provides a ready product with which to measure the multitudes of products offered him these days.

The preparation of a good specification is not easy. To detail a specification, such as one including detail drawings, assumes "cheap competition"—competition which aims only to meet the minimum detail requirements at a minimum of expenditure, regardless of the secondary standard functioning of the product or material. Too general or loose a specification introduces opportunities for inferior products to slide in, while the really desired and meritorious product loses out because it cannot compete in price with the inferior one. Writing good specifications is difficult, and those charged with this responsibility deserve the wholehearted support of responsible manufacturers of parts, materials and accessories, not only for the good they do in putting competition in these contributing industries on a practical and ethical plane, but for the assurance to the aircraft industry as a whole that reliable products are to be had by following the governmental lead in defining them.

### You're Wrong, Mr. Brisbane

REGARDLESS of all that has been said and written about commercial flying being a business and not a game, a large part of the public still believes that the transport pilot is a man who craves the thrill of landing against the elements; a man with a preposterous plan of adventure and daring in his eye; and a man who would chance all rather than be called a quitter by his comrades.

As an example of the above we borrow the following lines from the poet Arthur Brisbane:

"A pilot with youth's dream of being called 'yellow' is inclined to risk any weather. At every airport where passengers depart, the question to fly or not to fly should be left to some solemn person, not a pilot, and one influenced entirely by the passengers' safety."

We can only help but feel that Mr. Brisbane's association with transport pilots has been very limited indeed. Although we have no figures available at this time, we feel quite sure that the average age of the pilots employed on transport lines is well over thirty years. We also harbor the guess that at least sixty per cent of them are married men with children.

To those men the job of flying passengers from one point to another is a means by which they can provide for the wife and children at home. They take their jobs with just as much seriousness as the married man who commands the ocean grayhound, or the man who holds the throttle of the "Lanark." These their ability and judgment rest not only the lives of their passengers, and their own life as well, but the future welfare of those who are sent and their families. This is a team of many transport pilots among our circle of friends, but as far as we know, not one has a desire to leave a weaker and unhelpful to the charity of friends and relatives.

Metaphors is a subject that is included in the transport pilot's training, and although he may not become expert in this work he at least learns how to read a weather map. Once acquainted with what is ahead, the pilot is in the best position to judge whether he should go through or not. The "sensible person" as Mr. Brisbane terms him, may be officially responsible for the plane and passengers, but the pilot is doubly responsible for the plane, the lives of the passengers, his own life, and the continued happiness and welfare of his family. Refusing to risk bad weather, or running luck, are no evidence of a yellow stain in transport flying. On the contrary they are proofs of sound judgment.

For Mr. Brisbane, some are correct—and not for the first time. The American transport pilot is very much of a "sensible person," and personally we would much rather trust our necks on his judgment than on the judgment of all of Mr. Brisbane's solemn persons. . . . "Influenced entirely by the passengers' safety."

# Intermediate LANDING

By F. C. HINGSBURG  
Chief Engineer, Airways Division, Aeronautics Branch,  
Department of Commerce

**A**N AIRWAY is a navigable air space over a route provided with air navigation facilities. The essential difference between cross country flying and airway flying on scheduled operations on a safe airway is the ground facilities provided along the route to enable the flight to be continued under conditions of poor visibility by radio direction and communications or to provide for safe landing to avoid better weather. A daytime airway should be provided with suitable intermediate landing fields and current weather information made available to the pilot by radio. For night flying the airway should be equipped with boundary lighted intermediate fields, runway beacon lights, radio range beacons and current weather data compiled from observations along the route and made available to the pilot by radio.

Air navigation facilities on civil airways require intermediate landing fields approximately 30 miles apart to provide for the landing of airplanes under conditions of stress of weather or in the event of mechanical difficulties. The national airways are provided with landing fields on this basis. Under the Air Commerce Act appropriations for air navigation facilities are made available to establish, operate and maintain along civil airways all the necessary air navigation facilities except airports. Airports located along the route are considered to provide the necessary landing facility, and intermediate landing fields are only established where landing facilities and airports are non-existent.

Intermediate landing fields established and maintained by the Federal Government are occupied under the terms of a license providing for the right to carry out construction work incidental to the establishment, preservation and operation of the landing field, the right of the aeronautics public to use the facilities with rights of ingress and egress and other privileges consistent with the purpose for which the landing field is established. The license provides for occupancy for a term of years and renewable thereafter on a year to year basis. The license does not cover the proposed use of the field for purposes other than landing of airplanes under conditions of emergency and, therefore, the consent of the owner is required for the use of the field for purposes outside the scope of the terms of the license.

Other things being equal, the sites for the intermediate landing fields are selected in the vicinity of communities for the purpose of establishing co-operative landing fields. In accordance with the established practice of the Department which may, at some future time, develop into public airports. The community is expected, under these circumstances, to obtain a suitable tract of land by lease

*Although the construction and operation of the municipal airport may attract major attention, it should not be forgotten that intermediate landing fields play a most important part in the profitable operation of an airline. In this article Mr. Hingsburg relates in interesting detail of the requirements, establishment and operation of these spots of go-around, the right of which belongs cheer to the heart of the pilot who has to get down in a hurry.*

or purchase and to lease the site to the Government at a nominal consideration. The grading and reconditioning of the field is usually undertaken by the local community through its local government or local representative civic organization. With the establishment of intermediate landing fields on this basis, the Federal Government is able the necessary lighting equipment and maintain and operates the field under the intermediate landing field rules of the Department of Commerce until such time as the local community can take over the field and operate it as an airport.

Intermediate landing fields established by the Federal Government with or without the co-operation of the local community are intended for emergency use of the flying public. Regular and commercial use of the intermediate landing field by a fixed base operator is never permitted. A flying school course be maintained and permission must be granted in each instance for the use of the intermediate field for a limited period of time on specific dates provided it is shown that the use of the intermediate field is in the public interest. Permission for the use of intermediate landing fields will be granted subject to the approval of the property owners and the approval of the district office of the Airways Division responsible for the maintenance of the field and only in condition that the operator will place the field in the same condition in which it was found, the operator being liable for all damages incurred in connection with the use of the field.

## FIELDS

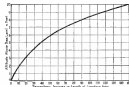


Fig. 1 Required Increase in Length of landing area at various slopes on level

Intermediate landing fields are located as close to the airway course as possible, although at many locations where landing fields are not available the fields are laid out on a straight line with the landing fields several miles on either side of the axis of the course. The landing field should preferably be on a highway or at least accessible to a good road to provide for emergency transportation. Locations are selected which are accessible to commercial power and telephone lines so that extension can be made at a reasonable cost to the intermediate landing field.

**T**HE IDEAL SHAPE of an intermediate landing field is square, triangular or circular, which permits landing in any direction against the wind, but such fields are not

usually found or available on routes passing over all types of terrain. Other shapes of intermediate fields are therefore used and an of the "L," "T" or "X" shape. The length of the runways at landing strips should be from 2,000 to 3,000 ft., especially in the direction of the prevailing winds and should preferably not be less than 500 ft. in width so as to accommodate the large airplanes now in use. At higher altitudes the size of the intermediate field should be increased in accordance with the curve in Fig. 1. Intermediate landing fields should preferably have clear approaches on all sides. An obstruction to the approach cuts down the effective length of runway from 7 to 10 ft. for each foot of height of the obstruction.

Intermediate landing fields should be as nearly level

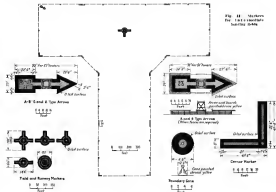


Fig. 2 Markings for intermediate landing fields

as practicable and have uniform surface without abrupt changes. The field should be free from ditches, rocks, dead stumps or any irregularities which would be hazardous to aircraft in landing. Natural drainage is of vital importance to prevent getting soft in wet weather. At certain seasons, sandy or sticky fields would present taking off after a hard landing. Intermediate landing fields are usually located in areas so as not to require artificial drainage as this is expensive. Landings on the intermediate fields are infrequent, and, therefore, there is usually no necessity for special surfacing. The ordinary crops of grass native to the particular vicinity are usually sufficient protection against the drifting of soil by wind and its washing by water. In many instances, if it is too expensive to scrape the surface periodically thus to establish a soil cover. It has never been found necessary to cut intermediate fields. This process is expensive and maintenance funds for doing the work in this manner have never been estimated and, therefore, not appropriated. The siling of a landing field will also be subject to the approval of the owner and the reactions of the lease as permanent injury to the land may result from such practice. It is inadvisable to plant landing fields which do not require some conditioning and surfacing. The average cost of plant intermediate landing fields is about \$1,000 per acre. In some instances, especially in mountainous locations, greater expenditures are incurred. The maximum cost of conditioning a field in connection with the Blue Canyon intermediate field located at an elevation of 5,300 feet in the Sierra Nevada Mountains on the west side of the slope, and the cost of cutting timber, removing stumps, grading and conditioning was approximately \$14,000. Fields without obstructions on the adjoining property should be selected wherever practicable, but where unavoidable, they should cover a minimum length of boundary and, if possible, be graded on one side leaving other approaches clear. Trees should be removed for a distance of about 500 ft. from the border of the field and the trees are purchased from the owner and removed by contract. There should be no high hills or bluffs as close to the field as to block approaches.

Colors and angles of the landing lights are installed to corner markers which are painted yellow. Black lines on the shape of the field and angles of the field for the pilot to determine without difficulty the boundaries of the prepared intermediate landing field. In order that landing fields may be easily picked up and identified from the air, ground markers are installed at the center of the field. Boundary markers, if within 30 ft. in length, are installed showing the direction of the best runways and landing strips. The directional arrow at the base of the tower points the direction of the runway course. Broken stone or gravel is used for the runway markers and the surface is sprayed with yellow paint. A border of black is obtained by spraying the ground surface with road oil confined to the shape as shown, and this prevents the growth of grass, making the markers more easily distinguishable from the air.

A wind cone is usually mounted on a bracket on the side of the airway tower showing the direction of the ground wind. At the end of the wind cone is a colored lantern, the light showing through the fabric of the wind cone. Intermediate fields are marked with a standard type airway beacon usually mounted on a 30 ft. skeleton tower. The boundary of the field is marked by white lights on low boundary standards spaced 20 to 300 ft.

apart, as located as to show the pattern of the field from the air at night. Green lights are required to show the best approaches to the landing field and red lights in the boundary lighting circuit indicate that there is no clear approach on that side of the boundary. Red lights are placed on all obstructions. Fifteen-watt lamps are used for white boundary lights and 25-watt lamps are used for colored lights. White boundary cones are used in conjunction with boundary standards as to prevent collision and define the field for day use. Reflective globes are used in connection with the boundary lights to increase the effectiveness of the boundary lighting system. Where seasonal power is not available, the boundary lighting system is operated on gasoline generators.

THE INTERMEDIATE landing field rules provide that all private pilots and the aeronautical public must comply with the Air Commerce Regulations and landing field rules promulgated by the Department of Commerce. The fields are established to provide safety of the aeronautical public in cases of emergency and no exclusive preferential privilege or commendation of any intermediate landing field will be permitted. An airplane making a landing has the right of way over airplanes moving on the ground or taking off. Upon landing the airplane shall be taxied off the landing area to a location that will not interfere with or become a hazard to an airplane that may find it necessary to make an emergency landing without warning. The take-off or landing shall not commence until there is no risk of collision with landing aircraft. No airplane shall be refueled while engine is running, and no engine shall be started under the pilot or other competent person shall be in the cockpit attending the controls and the usual system of signals and challenges understood by the operating personnel. Landings and take-offs where practicable shall be made into the wind and shall not be over dwellings, obstructions or parked cars unless unavoidable. In making test flights of aircraft subsequent to a forced landing, personnel making such flight will be limited to the minimum required to perform the required test. Unpowered aircraft shall be parked in the space allotted and shall be properly secured by ropes and stakes and when left unattended over night, shall be properly lighted to prevent collision by other aircraft. The caretaker employed by the Department of Commerce shall render assistance and control all accidents occurring on the field. All visiting pilots are required to report immediately to the caretaker to register and furnish information as to the cause of the landing and use of the field.

THE Department of Commerce has established 285 intermediate landing fields, the total acreage for landing field purposes totaling approximately 25,000 acres. This represents an average of 75 acres per landing field. The rental cost for landing fields is over \$100,000 per year, the average price per acre being about \$5. The total number of boundary lights maintained in the United States is approximately 7,000 which is an average of approximately 24 boundary lights for every field. There are usually five red lights marking obstructions on every landing field, visible at a range of about 100 obstructions lights maintained in the vicinity of landing fields by the Department. There are 120 lighted municipal and commercial airports along the highest airway routes in addition to the intermediate landing fields certified and operated by the Department of Commerce.

## THE Salesman's JOB

### A Few Timely Comments Regarding the Task That Confronts Those Who Hope to Sell Airplanes

By KARL E. VOELTER  
Carlin-Wright Flying Service  
Sales Promotions Manager

A YOUNG CHAP recently walked into our office and stated that he would like to sell airplanes. "Are you an airplane salesman?" was our first question.

"No," he replied, "but I can sell anything, and I certainly can sell airplanes."

Further questioning revealed that the young man never had been in an airplane; but, attracted to the industry by its sensational press, had decided without any preparation at all that his sales ability would meet with instant recognition and that he would be employed immediately (or so of the most important positions in the aircraft industry today).

If anyone takes as misguided as the hosts of young men who come to us seeking sales positions, and likewise possessed sufficient means to invest in an airplane, then airplane selling would simply be a matter of making deliveries at specified times, and the salesman would be a mere order taker. However, airplane selling requires far more than this. It calls for more, more, that the knowledge and ability possessed by the average "high pressure" automobile salesman who daily apply for sales positions with us.

It was he stated right here without reservation that the modern successful airplane salesman can only make sense of the fundamentals of flying, but also must be a psychologist of a high order. For the understanding of airplanes at present involves the merchandising of psychology to its fullest extent. In other words, the salesman must sell aviation as general to the prospect—must make him air-minded—before he branches into the task of selling any particular airplane.

To this end the larger manufacturing and distributing companies, such as the Carlin-Wright group, already have spent many thousands of dollars and know they will have to spend many thousands more selling aviation and building up an air-mindedness before they can attain the ideal of "just selling airplanes" in the routine manner that the automobile dealer sells cars.

Naturally, then, the problem of selling airplanes starts with the conversion of the general public into an air-minded public, and in this direction we are accomplishing results. The Carlin-Wright Corporation and other large

manufacturing and distributing groups have constantly encouraged their officials to speak at luncheon clubs and at formal and informal gatherings of all descriptions, knowing that, although they could expect an immediate response from such efforts, they were laying the cornerstone of future sales not only for themselves but for the industry in general.

The Carlin-Wright Flying Service, a unit of the Carlin-Wright Corporation, maintains an airplane showroom at 22 West 57th St., New York City. This showroom was opened as a sales laboratory, and its institutional value is much above its income producing possibilities. The salesman on the floor are selling airplanes, as well as accessories, flying school courses, charter transportation and aerial photography, but one of their important functions has been to sell aviation to the people from every walk of life who enter.

Everyone engaged in this task knows that he is selling his competitors as well as himself, but that is part of the necessary work required of the industry.

Many of us remember the time when people would not ride in an automobile. Such an idea would be laughed at today. A similar thought, however, is voiced by persons who say they want to fly if they can lose one foot on the ground. If we can get these people through their first trip, the majority of them become flying enthusiasts. It has often been true that those whose doubts about flying are overcome by a demonstration of its comfort and safety become its strongest advocates. We regret that we are unable to afford airplane rides to all the world, but such promotion would imply giving rides to none.

Upon the industry and, therefore, upon our salesman falls the task of winning into others our own belief that we are selling transportation which has no peer. Before the average salesman can expect the airplane as "just another high priced commodity," the highly trained airplane salesman must first sell the idea of flying. Until the public becomes air-minded, the airplane salesman must possess more than general sales ability. The airplane salesman of today, with his specialized aviation training, must pave the way for the "high pressure" salesman of tomorrow.



increased in his own season he desired. Furthermore, that he has a greatly decreased chance of even becoming a pilot, in accordance with the seriousness of his defects, is evidenced by the following discussion of the relation of these standards to ability to become a pilot.

The numerous criticisms of the Physical Standards for Airplane Pilots as adopted by the Department of Commerce, having arisen, particularly as to their being too arbitrary, it was decided to correlate them with actual records in student pilots.

Our complete file was gone through by H.J.C. and the records of all those who had started in as students 12 months or more previously, or who obtained a pilot's license in less time, were studied. Thirteen months was adopted because a student permit is good for one year. One month was allowed in addition to cover the time necessary to get flight tested, etc. Only students were selected in order to eliminate the effect of previous training. The idea of the study was to determine what percentage of students with and without physical defects reached a higher grade. There are many factors determining this. For example, a student may lose interest, he may be unable to go on for financial reasons or he may be unable to learn to fly for other reasons. Inasmuch as all these factors affect all groups equally they may be designated as having no effect on the statistics compiled. A total of 9,103 records were studied. The only basis of selection was that continued above, namely, that they must have started as students at least 12 months previously, or have obtained a pilot's license in less time. All such cases were selected so that the study represents all material available. The number studied is sufficiently large to warrant drawing deductions.

**PHYSICAL EXAMS** were then classified as follows: The first group consisted of those students having no physical defects, of whom there were 7,662.

The second group was composed of those students having no physical defects, who eventually received a pilot's license, namely, transport, limited construction, limited or private. Of these there were 2,685.

In other words 38.4 per cent of all students without physical defects eventually obtained at least a private license.

In the third group those students having minor physical defects, which, while disqualifying for transport (Pilot), are not disqualifying for private or limited license. Such defects as 20/30 vision, slight defects of the muscular system, minor defects of the ear, nose and throat and color vision, etc. These defects, prior to this study, had not been considered by us as having any bearing on an individual's learning to fly, but as fact of future trouble or causes of dissatisfaction arising later. In this group there were 599 cases. Of these 303 eventually reached at least a private grade, or 50.3 per cent. In this group we included minor defects of color vision. The results of this study have suggested to us that minor defects of color vision may be of no consequence and hence these cases should perhaps appear in the first group. However, inasmuch as it is a minor defect, these cases have been left in the second group. This group demonstrates that in spite of these being minor defects, they apparently decreased the individual's chances of getting even a private license, even though they were not disqualifying for a private license.

The fourth group was composed of those having major

physical defects still qualifying for a private license, but disqualifying for a passenger-carrying grade. Defects such as 20/40 or 20/50 vision, gross defects of the ear, nose and throat, slight restrictions of function in the extremities, certain minor disorders of the nervous system, not serious enough to disqualify, were among those listed. This group it was expected would have more difficulty in learning to fly and the results of the study confirmed this. There were 404 cases in this group and 73 eventually received at least a private license or only 18.3 per cent.

The fifth group consisted of those who had defects that were disqualifying for all grades, but who for some reason were allowed to fly. Usually, the reason was because the examining physician issued him a permit in error and so the basis of it the man spent considerable money before he was notified of his disqualification. In such cases the man was given an opportunity to get a private license. Yet of 40 cases in this group only 5 received a private license or better, and one of these was killed almost immediately after receiving a private license. The cause of his fatal accident was unquestionably due to his physical condition. These really only four cases ever got anywhere or a total of 10.0 per cent.

**DEFECTS** were 18 months or a year on as the basis of a higher percentage would qualify these licenses, but this would affect all groups equally and there is no reason to suppose that the ratio between the groups would be in any way altered.

As a matter of fact those in the last group had fewer failures, other than physical, operating against them than did the other groups. They were all men known to be intensely interested in flying and who actually began training, so none of them dropped out because of lack of interest or presumably for lack of finances. Hence we might expect a higher percentage here and yet it is distinctly lower.

From this study of over 9,000 cases it certainly would appear beyond question that a man's physical condition is a direct reflection on his ability to learn to fly. Hence it would also appear that a man is taking a long chance in starting training unless he has passed the roughest physical examination. If he barely passes the private grade, his prospects of getting anywhere are almost cut in half. If he cannot pass the private grade it would be useless to consider him for a career as he has not over one chance in ten of ever progressing.

**THIS STUDY** has in our opinion vindicated the standard set by the Department as not being too strict and further indicated that the standard for private grade is probably too lax. It has not further supported our belief that writers should never be granted students. While these will always be individual exceptions to any given rule, this study conclusively indicates that as a group, men with physical defects have a chance of becoming pilots that stands directly in contrast with the seriousness of their defects.

We have decided that a further study is warranted with a view to investigating the effect of specific physical defects on flying ability, rather than physical defects as a group. Certain defects may be of no consequence and tend to raise the percentages. Others may be more serious and tend to lower them. The study will be made in the course of the year as we do not then have available about 30,000 cases instead of 5,000.

## Design IN 1930

### Concluding the Symposium by Leaders in the Industry on 1930 Airplane and Engine Design Trends

**I**N ORDER to provide an indication of the design trends in the aviation industry during the coming year, questionnaires were sent out by AVIATION to leading executives and engineers in aircraft and aircraft engine manufacturing companies, as well as to a number of other recognized authorities. We were fortunate enough to receive many more answers than we had space for in the special issue of February 15, devoted to the St. Louis International Exposition. Many were published the following week. The remainder appear hereafter.

The following questions, given in slightly abbreviated form, were asked leading engineers in charge of airplane design:

1. What is the general trend of design likely to be and in what phase is greatest progress likely to be made?
2. How much further evolution is to be expected in the design of very large planes, and what prospects are there for immediate production of larger planes?

#### Thick Airfoil Sections With Smaller Center of Pressure Travel—Superchargers—Magnesium Alloys

By JEAN FRADDES

Senior Engineer, Fabry Aircraft Corp.

**T**HERE WILL PROBABLY be no important improvement in the structural and aerodynamic construction, due to the use of better materials, especially in the line of light alloys, nor double the steps with look somewhat similar, design, understanding that something which does not look good is not good, but the greatest improvements to come are certainly of an aerodynamic nature, because this science is yet so little known that anything is to be expected.

Of course, the development of aviation in the near future will also be a direct function of the progress of the power plant.

I do not think that responsible companies will design much larger ships than the last creations—not because

3. How much increase is there likely to be in the use of aerodynamic novelties and striking departures from accepted form, such as the slotted wing?

4. Which way will the weight of armor swing in the controversy between confidence and economy based wing structures?

5. Will wing loading continue to rise?

6. Are aspect ratios likely to increase, decrease, or remain approximately the same, on the average?

7. Will metal construction tend to replace wood, and if so how rapidly?

8. Will metal construction continue to be predominantly of duralumin, or will alloy steel step into favor?

9. What developments are to be looked for in the near future in landing gear design?

of difficulties in realization, but on account of economic problems yet unsolved.

As far as aerodynamic novelties are concerned, the field is so immense and so uncertain that it does not seem wise to express an opinion.

4. The aerodynamic type of construction has decidedly won the battle against the multiple, and now the conventional type will give place with the public completely.

The actual thick wings do not present any advantages as far as weight is concerned, but they have as good aerodynamic characteristics as thin wings and are far more efficient, and the faster thick wings will be even better as the airfoil design is perfected.

Thick tapered wings will be designed with a very small center of pressure travel and a small moment coefficient, thus increasing their inherent stability and reducing their deflections, especially when rigid covering is used. Ships equipped with such wings will need but small tail surfaces.

5. The unit wing loading will probably increase to give speed while same device will be provided to reduce



loading speed. Wings with variable area and center are likely to be perfect.

The truly important improvement which will increase the speed with the present type of equipment will probably consist of flying faster to fly fast. Supercharged, over-compressed engine with automatically adjustable pitch propellers, perhaps retractable gear with several speeds, and variable wing area may solve the problem and revolutionize air navigation, in which the radio will play a dominant part.

6. Change in aspect ratio is intimately connected with the improvement of the materials and for the wing construction.

7. Both metal and wood construction are likely to be used and improved, as up to now neither type has proved decisively superior to the other. It seems to depend upon the particular case.

8. It is almost certain that the use of duralumin as compared to alloy steel strip will continue to be more widespread, for airplanes constructed in the near future. The discovery of light alloys with high mechanical properties is to be accompanied as research in that direction is being carried out in many European laboratories.

We may also confidently look forward to the perfection and perfection against corrosion of the present magnesium base alloy. In either case the improvement would replace duralumin because it has a much better strength/weight ratio and is very easy to work.

9. Further progress of also air-cooled, or also spring shock absorbers as the expense of the roller road type is to be expected during 1950.

• • •

## Work for Safety—Research on Ultra-Light Alloys

By CHARLES HEALY DAY

Chief Engineer, Chief Engineer, Air Transport Aircraft Corp.

THE CHIEF PROBLEM now confronting the manufacturer and the engineer is to supply the consumer with the product which the consumer's experience leads him to demand. Undoubtedly, the engineer dictated the need of design according to his own

disposal ideas. The airplane, after all, is not another and improved method of transportation and its body and will be taken out of the spectacular, novel and super-charge class.

As all other important transportation vehicles have been perfected through the process of expansion, so, also, must the airplane be developed primarily by the operator and, second, the engineer being relegated to his rightful position of finding ways and means of supplying what is most needed rather than developing novelties which may or may not be useful.

No matter how optimistically so many view the present airplane transportation situation, that which holds back the acceptance of air transportation by the public still is what it always has been—fear.

Safety of operation is always in the predominant place, and until now frequent landing flights are available safe operations is possible only with airplanes having a rela-

tively low landing speed and an excess of stability and controllability.

The public will never consider a means of transportation entirely safe in which either failure occurs possibly without injury or death. Take away anything that can be spectacular, and how many of us as a whole would frequent trips of hundreds of miles in an automobile if, while at it, a minor failure meant running the danger of serious injury or death?

The public is demanding safety. Therefore, it is to be expected that the general trend of design will be toward greater structural strength, a disposition of loads which will cause the least injury to occupants in case of crash, greatly improved stability, and more effective controls.

Improved performance will come with clearance of detail, increased aspect ratios, and in some types by the use of highly tapered cantilever wings, other than by the sacrifice of landing speed through increased wing loading.

In almost any of the existing types of airplanes, a relatively enormous increase of stability and controllability is possible without the use of auxiliary devices of non load and resistant detuning materially from efficiency, likely or all-around performance.

Also, with the present knowledge there is no excuse for the design and construction of any airplane which has had serious shortcomings. It is easily possible to build auxiliary devices, to manufacture airplanes which cannot be spun, but it is believed this would be undesirable because of the suspicion that such planes might spin without recovering.

Improved stability and controllability will be combined with wider wheel trends and wider landing gear. It is regrettable that many airplanes, because of lack of these essentials, are incapable of getting safely into landing areas which are provided by their wing loading. An improvement in this area is necessary.

Development of aerodynamic perfection will continue, of course, but until the present types of airplanes are more safely perfected according to the knowledge which has been available to all engineers for years, the use of such novelties is not justified. The development of the shod wing and auxiliary devices is to be discouraged, but not as an attempt to improve an inherently poor design.

Because of lack of experience with the larger of the existing types of airplanes, it is doubtful if responsible manufacturers will in the immediate future attempt airplanes of much greater capacity than those now exist.

The use of metal in airplane construction must continue to increase. Although it is true for all parts of some types is not now justified structural strength and production economies demand that eventually practically all airplanes will be of metal. The light metal alloys will revolutionize over steel because of the better form factor available. The development of magnesium alloys and perhaps beryllium alloys has interesting possibilities for the future.

To summarize: (1) The engineer of tomorrow will design into the products desired, not what he personally likes; (2) he will put more stress on aerodynamic; (3) he will improve performance by improving design, rather than by sacrificing low landing speed; (4) he will experiment with novelties but will not fall back upon them as accepting to improve inherently poor designs; (5) he will move carefully in increasing the use of light metal alloys; (6) he will gradually increase light metal alloy construction because of strength of structure and economy of production desired.

## Higher Aspect Ratios for Stability—Effects of the Airbrake

By ALBERT S. HEINRICH

Chief Engineer, Aeromarine-Kelvin Corp.

KNOWING COMPETITORS will force designers and manufacturers to give more attention to aerodynamics and there will be a general trend toward cleaner and less wasteful design in order to obtain better performance. Many nations will be given to structural design in order to produce structures of lighter weight and more uniform strength throughout the entire structure. There will be a general tendency to favor airplanes designed by competent, experienced engineers and designers and built by responsible manufacturers and there will be less encouragement given to "theorized" engineering and inexperienced designers.

Manufacturers will do their best to serve as a good barometer to indicate to what extent the traveling public will take to the air as a means of transportation. Encouraging transportation agencies will serve to ensure responsible manufacturers to produce larger passenger carrying airplanes in order to reduce operating expenses of the transportation lines.

Experimental development of variable lift wings will probably be carried further this year than ever before, since it is most evident, as the present application of the airbrake as a means of the transportation is to be used to offset that a high top speed is necessary in order to maintain schedules against strong head winds and still planes must have sufficiently low landing speeds to make a landing safely in a small field in case of necessity.

There is still no general trend toward either full cantilever or the externally braced wings, which seem to meet with equal favor among the designers.

Unit wing loadings are as high now as they will go for some time on commercial airplanes, unless higher lift wings are developed, since many crashes are due solely to high loadings. Increasing wing span extends outward at the expense of bending speed.

Increased demand for stable airplanes will no doubt tend to influence the trend of design toward somewhat larger aspect ratios.

But construction will be greatly weakened, particularly by the more experimental manufacturers. I look forward to a much greater use of heat-treated steel structures. The use of duralumin and other aluminum alloys will continue to increase in airplane structures and the demand for these alloys will no doubt be effective in lowering the price of metal in products as normal.

The use of airbrakes with modified shock absorbers will be conducive to better landings and tend to induce stress in the structure. Somewhat wider trends will no doubt be favored by some designers and there will probably be a general tendency toward low wing configurations, particularly in the light plane class, with greater effort to keep the vertical position of the C.G. as low as possible to reduce the possibility of stalling over.

Accidents due to faulty design, both structurally and aerodynamically, will be gradually lessened as there will undoubtedly be a tendency toward a closer check on all designs and structures, even in experimental airplanes. The aircraft industry cannot continue in a healthy condition unless the inexperienced element is gradually eliminated.

The weight of public favor will only be shown in the support of the airplane as a means of transportation after

consistent performance and reduction of accidents has proven its worth and instilled confidence in it as a fast, reliable and safe means of travel.

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## Safety and High Performance—Go Slow on Metal

By WILLIAM WAIT, JR.

Chief Engineer, Babcock & Wilcox Inc.

IT IS CONSIDERABLE the need of aircraft design it becomes apparent that it is not reasonable to expect anything in the near future that will materially change the general concept of the airplane as it exists today.

It is hardly probable that we shall see a considerable increase in the number of airplanes designed and built to the D.O.C. as apparently large boats of this type are

the only ones which have any hope of survival if forced down on rough water. It is possible to reduce their heat drains as airplanes since take-off and landing areas are more or less unrestricted, but it is not reasonable to expect to see their equal in size as landplanes, due to the landing and take-off problems involved.

The use of metal in the construction of the airplane type plane is receiving a great deal of attention and it is to be expected that an increasingly large number of planes of this type will be offered to the public.

We may expect a steady and easy or less rapid advance in all design toward the increase of safety of flight. These are necessary in order to provide the increased market which we must have if aviation is to progress, and it is not progress if we, in the aircraft industry, are to survive.

The rapid increase in the number of planes flying daily at all airports is placing an increasing pressure on the ability to see in all directions and will require a considerable increase in the vision afforded by most of the present instrument types.

It is probable that any great increase may be expected in the number of planes equipped with shock wings, due to the involved situation which exists as regards the present rights.

The development of the folding aileron, due entirely to the findings of one of our country's best known aerodynamic experts, has not probably the most satisfactory result of the Guggenheim Safe Aircraft Contest and its use may shortly be expected to become more or less general where slow controlled flight is of such paramount importance as it is in passenger transport.

There seems to be a tendency to get away from the very small light type of airplane which seemed very much on the increase a year ago. The tremendous industry went through the same experience with the cycle-car is important to provide high performance with any vehicle of transport if it is to become popular, and while there is a great deal of power involved in the use of the airplane, the performance becomes necessary.

Structural modification such as the increased use of steel throughout the entire airplane is to be expected. The use of steel is particularly becoming increasingly important because of the rapid exhaustion of the steel.



Charles H. Day



Wm. Wait, Jr.

able wood supply. It is, however, considerably limited at present because of the expense involved in the build-up structures required where any attempt is made to keep weight at a minimum. The use of duralumin in particular involves a great many difficulties, and it is to be expected that metals or alloys will be evolved which will make possible the use of modern automatic welding processes or other methods which will eliminate the highly expensive repetitive hand operations now necessary.

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### Still Higher Performance and Easier Maintenance

By JOSEPH S. NEWELL

*Act. Professor of Structural Engineering, Massachusetts  
Institute of Technology*

PERHAPS THE GREATEST PROGRESS in aeronautical material during 1929 has been occasioned by the almost universal demand for higher speeds and increased performance. As evidenced by the widespread use of streamlines reducing drag and by the general desiring of very exposed details, designers have obviously made a definite effort to utilize to their utmost all improvements which enhance performance and permit the full utilization of the increases in power and efficiency made available by the engine advances. In these efforts they have achieved a considerable success and it is to be expected that 1930 will see still further advances.

It is also anticipated that more thought will be given to the design of these details which require maintenance in service. At the present time an inordinate proportion of the costs of operating commercial aircraft accrue from the overhauling, adjusting, or replacing of small parts which are subjected to wear. Such costs must be eliminated where possible and this coming year should show definite results from the efforts to remove or reduce them.

The large production increases last year and followed during the past year demonstrate the durability of all-metal construction for airplanes. Moreover, there will undoubtedly be definite progress toward the solution of the engineering problems inherent in metal wing and landing designs. It is to be expected that the most competent of its engineering departments engaged in research programs, the industry will advance far toward the ultimate solution of these problems. The construction and performance criteria obtained by many manufacturers to determine facts and to obtain data which will permit the introduction of methods of analysis and design are indicative of the sound development to be expected. It is also apparent that the more progressive designers are not contenting themselves to all-metal projects depending upon the inadequate and in many cases, extremely fallacious data which are available at present without subsequent efforts to obtain structural factors by static tests of entire airplanes.

The present year should, therefore, bring about a sound development of all types of wings with a trend toward those types which are inherently clean in design and from which high performance may be obtained with minimum operating costs. The Guggenheim Safe Aircraft Corporation having established the fallacy of attempting to accomplish the impossible, it is not unexpected that the most careful designers will design very far from the present types of wing and fuselage structure.

### Progress Along Conservative Lines

By THOMAS CARROLL

*Technical Advisor, Federal Aviation Corp.*

DEVELOPMENTS in the aviation industry during the year of 1930 will be largely developments from the newness which have been created by the very thorough shake-down which the industry has had during 1929. It is already beginning to appear that some of the innovations during the past year, which have been almost purely in the nature of newness, will in the long run prove a benefit. Nobody will demand a greater usefulness and productivity.

Usefulness in the preponderant feature of aviation which must be stressed right now. Every activity in building, design, manufacture and operation, must be carefully considered from the point of usefulness.

For any particular assignment, whether in transport, schools, pleasure or sight-seeing, it is obvious that a specification can be laid down for an airplane which will best perform that function and for which a certain extent or compensated type, which is not beyond sound aeronautical engineering, can be provided.

As airplane from now on can be, and I believe will be, required to possess maximum attainable efficiency for its job. And further, two items which have in the past been not so carefully considered will be of importance. Speed is a need that eagerness to attain even at the expense of structural weight.

Development of better airplanes cannot look to improved fundamental aerodynamic design in the immediate future. Production is now at the birth of research, which inevitably should be two or three years in advance. That at the present time of aerial design is quite necessary in current production.

Supplemental devices for aerodynamic improvement, slots, flaps and other such devices have their place and will be used. I believe, however, that their place is more definitely in the use of equipment that are developed, i.e., for training and building up experience. Beyond that, their place is not indicated in the practical work of operators or pilots.

Wing loading has probably reached its peak, and the tendency will be to fall off in this respect. Airplanes must come from the fields and not from the fields we like to visualize. Established airports within reasonable proximity to large centers of population, particularly in the East, are not available in the expense area to be at frequently recommended. Aspects of more limited type, but which are logical and useful, are beginning to be available, and the airplane must adapt itself to this condition rather than follow the tendency of the past to suggest sheer geography. Therefore, low wing loading, with attendant shorter take-off and landing climb, which I consider of more importance than short landing, must be contemplated.

Aspect ratio, previously considered a very definite advantage, is no longer attractive. Some airplanes with very low aspect ratios have proved efficient in the air, and there can be no argument that more compact dimensions, particularly in regard to span, are of very great importance in ground handling and storage.

Perhaps the present field for development is dependent upon the perfection of materials which are particularly adaptable to aviation uses. Metal construction is most attractive from the point of view that it appears to be already demonstrated that all-metal airplanes can be built in which maintenance and depreciation have been reduced

to a minimum figure, but the metals are not entirely satisfactory. Their greatest value appears to be largely on paper from their low weight for strength ratio. Unfortunately, however, the strength factor depreciates rapidly due to corrosion and general fatigue. It is regrettable that steel wood veneer and fiber wing covering are not to be replaced in the near future.

Altogether, in summarizing, it seems that the development of equipment during the year 1930, and very possibly extending further into the future, will be accomplished by taking very careful stock of the available features of design, preventing them in the most favorable combination and thereby providing airplanes for individual projects which are to the highest possible degree useful.

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### Giant Airplanes—Go Slow on Abandoning #ood

By FRANK M. SMITH

*Superintendent, North American Corp.*

I BELIEVE THAT the general trend of design will be toward the low-wing monoplane, as it is planes of this type which most designers feel have the greatest future.

I believe that there will be some very interesting developments in the line of large planes in the near future. Such planes will be built for both cargo and passenger and far better accommodations for passengers.

I have in mind a design which I have recently seen completed by a nationally known designer which is even equipped with train beds, car porter and smoking room. This plane is designed to carry 45 passengers.

There will without doubt be a large amount of development along lines of stream wings and safety devices as a result of the Tanager success in the Guggenheim competition. Already many designs are under way embodying a great many of the features used on the Tanager.

The controversy between the cantilever and externally braced wing truss exposures is due to continue for some time, but I believe that the cantilever wing will be the wing of the future, brought about by use of material with better strength factors than is now obtainable.

I believe that the trend toward this rise in the wing loading has about reached its limit and that there will be little advance in this feature in the future.

I believe that aspect ratios are likely to remain approximately constant on the average. These are, however, being changed by those working toward an increase in aspect ratios at the present time.

Undoubtedly there will be further progress along the line of elimination of wood in airplane structure. It is questionable, however, if such a policy is sound at the present time as regards expense in manufacture. It is surprising but no less true that although a wooden plane was the first we constructed, economical use of wood in airplane manufacture is to be found in practically all of the factories in the United States. There is no question but that at the present time proper investigation into the problems and use of wood will result in the manufacture of wings at a price which cannot be approached at the usual construction at its present state of development.

I believe that the use of duralumin in construction will be replaced in the next two or three years by the use of aluminum alloy and even possibly by the use of steel.

I believe that the introduction of the Maestrius

air wheel will have great effect on landing gear design and that all planes in the near future will be equipped with brakes, and steerable tail wheels.

The tendency to nose over is largely overcome by the use of the air wheel, and its landing gear can be made as solid ground without danger.

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### Better Landing Gears and Multi- Engined Planes

By JOSEPH KREITER

*Chairman of Board, Joseph Kreiter Corporation*

IT IS NOT POSSIBLE any noticeable change will be made in structural or aerodynamic design, but more attention will be given to clearance of details.

There is probably no change to be expected in large planes for the present year due to economic conditions of the industry. Volume of air traffic will be the large governing factor in regard to rate of planes.

I believe manufacturers will stick to standard design, although extensive experimental progress will go on in all branches.

Cantilever braced wing construction will continue for reasons such as cost in production and replacement. Metal construction will favor cantilever methods.

Wing loading will remain practically constant for the time being.

Aspect ratio should remain practically constant. Metal construction will increase gradually, particularly during the present year.

Duralumin structures will continue to preponderate. Shock absorber travel will increase for softer landing. Landing gears will undoubtedly have wider travel due to economic advantages gained. Modification will be required on light plane brakes to prevent nosing over.

I believe trend will be more toward multi-engined planes to a factor of safety—moving from 4-plane jobs upward.

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### Higher Cruising Speeds Without Increase of Landing Speed

By FELIX W. PAULOWSKI

*Department Professor of Aeronautical Engineering,  
University of Michigan*

IN ORDER to attain that I have certain one or less definite points of view upon the various questions presented. However, I am not quite sure that I shall be able to answer them specifically as to the time limit (1930-1931), since some of the developments may take a while of years for realization, and it is impossible to foresee what part of the developments will be accomplished within the year 1930. It is also difficult to separate in one's mind the things that one would like to see accomplished from the things that one has accomplished under circumstances governing the development of a new art and industry in a country.

This question is very intimately connected with another question, in my opinion perhaps the most vital question in connection with the future of commercial flying in this country, namely, the question of increase of cruising speeds without increasing the landing speed.

Present cruising speeds (around 90 mph) is certainly too small. At these speeds the time saving element is very small in comparison with what railroads and automobiles can offer, and the situation is still further aggravated by conscious detours or poor transportation between airports and business centers of many cities.

This situation is not so acute in Europe, due to the much lower costs per mile (which, of course, are possible there, thanks to the government subsidies) (the relative condition has been changed, since Professor Dornier's statement, by the sharp reduction of rates on most American lines—Ed.). Besides, the European airports have, as a rule, somewhat higher cruising speeds. (American operators will strongly question this—Ed.) If the American public could be offered cruising speeds of 150 mph, or better, it would undoubtedly increase greatly the percentage of our air travel. It seems to me that well-aided an energetic and successful effort on the part of our airplane manufacturers is that direction, the aerodynamic industries will suffer none a severe effect, which may impair the future of aerial transportation.

It would not do to increase the cruising speeds by increasing the wing loading and the power loading of airplanes, as this would result in a parallel increase in landing speeds. While landing, even at 100 mph and higher, might be perfectly safe as good airports we have to count with emergency landings on terrain at which such landing speeds are prohibitive.

It is, however, possible to increase the landing speeds without increasing the landing speeds, although it is a rather difficult problem, requiring numerous trials and careful and delicate adjustment of the various aerodynamic elements in the design. A good example of the feasibility of this are the French propeller planes with two small speeds of about 100 mph and cruising speeds of 30 mph, then having speed ratios of 4-1 in monoplanes and biplanes as well, and this, without any artificial such as flap doors, variable cambers or wing areas, or retractable leading gear. The speed ratios, with perhaps one exception, are only between 3:1 and 3½:1 in our commercial planes.

2. At the present stage of development of air travel, I do not see any immediate necessity for larger planes, i.e., planes carrying 20, 30 or 100 more passengers. As I indicated, in Europe airplanes are more on the average, with one-third of the passenger load for which they are built. While the problem of design and construction of large airplanes is most interesting for the aviation engineer, the over careful manufacturer will very likely obtain from producing such airplanes, being paid with the present market for sub-planes of 5, 10, 15-passenger capacity.

3. Staking airplanes from accepted forms, such as the dotted wing will very likely not appear as commercial airplanes in the immediate future. The dotted wing certainly presents very interesting possibilities. However, the whole question of dotted wings is still very little known, for instance, there seems to be no facilities for slotting the wing in several different ways for different effects or purposes, and this problem may branch out into that of removal of the boundary layer. But while all these problems will furnish plenty of food for activities in aeronautics, they will not, I think, furnish sufficiently reliable results for the designer to base his commercial work upon.

It seems to me, however, that the rather moderate addition of variable camber by means of flaps (rear, or front and rear) offers considerable aerodynamic advantages in

spite of structural complications, and since the experimental study of an individual case does not go nearly beyond the routine wind tunnel test in order to secure the necessary design data, I therefore expect airplanes of such a type to appear in the near future.

The next thing may be said in favor of retractable leading gear particularly in connection with low-wing airplanes, which lend themselves to this improvement without undue structural complications and increase of weight.

I venture to suggest, if not to predict, the appearance also, of the "flying wing" type airplane; not the type one as dreamed of by Professor Fokker, since the time for it is not quite ripe, but the small and middle size commercial "flying wing" which equally well offers considerable aerodynamic and, what is more important, structural advantages and economy of operation.

4. This question is most intimately connected with that of use of load capacity. It looks as if the limit of use in pure cantilever and in semi-cantilever has been reached already (considering the strength and lightness of the present materials) in the last large Junkers biplane and Dornier's DO-X seaplane, respectively, although the latter could be made properly revised, as seaplane. Since I do not see any necessity for commercial airplanes of such sizes, therefore, the pure cantilever type for smaller and the semi-cantilever for middle size planes, both offering well-known aerodynamic advantages, present the best solution for the future, and will be favored by designers and manufacturers. It should be kept in mind, also, that it is easier to design an efficient cantilever than a biplane for these moderate load capacities. Usually, I do believe, however, that it is possible to build biplanes and monoplanes as efficient as monoplanes, but this will become a reality only when we are in position to build large planes carrying hundreds of passengers, something that belongs to a somewhat more distant future than 1950.

5. This question is very closely associated with that of increase of cruising speed, discussed under question 1. Since higher cruising speeds are at present impossible for the welfare of commercial aviation, progress in this direction will undoubtedly result in increase of wing loading. The successful solution of this problem will be facilitated by the to-be-expected increase of reliability of engines. The well established aircraft Otto-cycle engine industry will fight hard for existence in the face of a new "rival" in shape of the Diesel-engine-engine; besides, the Diesel-engine-engine is too new to compete in reliability with the gasoline engine, so that the latter can still believe it a good many years in which to hang its laurels. The designers and manufacturers of the present engines will put all of their accumulated experience, skill and energy behind their product and, therefore, we may expect in the near future a marked increase in the life and reliability of the gasoline engine, parallel with which will go an increase of cruising speeds, decrease of wing loading, elimination of emergency landings, and simplification of landing gear.

6. The aerodynamic advantages and the structural disadvantages and cost of high aspect ratio are by now well established, and known to designers and manufacturers. The present popular rules of thumb are, thus, being the result of compromises based on experience, will not likely be subjected to changes in the near future, and will remain within the present limits, which are sufficiently wide for design of reasonably common types.

7. In spite of all the advantages of metal planes above

those made of wood, primarily metal is far from the ideal material for airplanes, and wood construction is not displaced of all its merits. For example, as I understand, in European commercial flying, the service life of a metal plane is accepted at five years and that of a wooden one at three years, while the cost of a metal plane is about twice the cost of an equivalent wood construction. The operator can therefore derive much service at the same money outlay from two wood planes than from a metal plane, and so the wood plane will continue in existence for some years to come.

8. Some very interesting applications of steel strip construction have been developed in France and particularly in England. We have also, very rarely, examples of drilled use of steel tubes in wing structure.

**P**ROMINENT aircraft engine designers were asked independently the following questions:

1. Is the liquid-cooled engine likely to regain substantial favor during the coming years?
2. Will air-cooled engines continue to be predominantly radial in form?
3. What are the most important improvements in engine performance to be anticipated in the near future?
4. Do you think that it will be possible to make further increases in compression ratio without change of basic fuel?
5. Are you in favor of a marked improvement in standard fuel as governed by specifications?
6. What is your opinion of the probable early future

### **Radials, Air-Cooled, for Dependability— Controllable-Pitch Propellers— Magnesium Soon**

By E. K. WILSON

President, Hamilton Standard Propeller Corp.

**I**n my opinion, the liquid-cooled engine is not likely to regain substantial favor during the coming year. I believe the general trend will continue towards air-cooling. While classical cooling appears to offer certain advantages over water cooling, nevertheless, any direct cooling process must be inferior to the direct cooling process from the point of view of dependability. As time goes on, we begin to appreciate more and more the importance of dependability. The whole progress in an airplane is almost directly dependent upon this factor. People will begin to fly with experience, has demonstrated conclusively that air transport is as dependable as any other form. Certain transport companies have demonstrated that it can be made more dependable. In my opinion, we can afford to sacrifice

this economy. I would not be in the least surprised to see an increase in use of steel for wing construction in the near future, although this might depend upon to a considerable degree upon the policies of our domestic producers.

9. The successful development of the Mousseline type of fire, and experience in the use of it, may result in reducing the necessity of larger radii of the most shock-absorbing devices. In general, the whole question of landing gear design has very much to do with the merits of our landing fields and with the necessity of allowing for emergency landings on not quite suitable terrain. The changes in landing gear design will, therefore, have to go parallel with the improvement of landing grounds and with the progress in reliability of the engines.

of the compression-type in aircraft engines.

7. Do you look forward to marked improvements in cylinder arrangements?
8. What is the prospect for the use of overhead intake the ordinary present stage of commercial practice such as magnesium, and of new manufacturing processes, such as the forging of cylinder heads of aluminum?
9. Are you satisfied with the present type of engine mounting and the ways in which it is applied?
10. Are any marked improvements in necessary installations and in "planning" to be looked for?
11. Will there be a trend towards the building of full power plant sections, with the engine designer providing for standard mountings, cowlings and accessory locations?

a good first one weight, first one, economy of operation and dependability, to double dependability. In my opinion, air-cooled engines will continue predominantly radial. I do not anticipate a trend towards the development of in-line, V or X form engines. In most of the modern engines, low cost, light weight, ease of maintenance, durability and dependability are directly due to the single-row radial form. The only advantage to be had in another form of engine is the possible advantage of improved vision. In my opinion, the price paid for this improvement is too great, as it involves departure from the single-row radial form.

The most important improvements in engine performance to be anticipated, should result from the use of higher compression ratios, supercharging, controllable pitch propellers and reduction gearing. All of these are more or less linked up with improvements in propeller design. There is a large field open for improvement in this line. Lighter propellers may be obtained to permit the use of reduction gears. Controllable pitch propellers must be had to permit the use of supercharging and over compression, and the improvement of performance which will result from their use. I believe substantially

reduction in weight can be had in the power plant as a whole, without any loss of dependability or durability.

Further improvement in power plant performance can be had through improvement in fuel. The diesel opportunity here lies in the development of anti-detonating fuels. Perhaps along this line may be found the simplest, easiest and cheapest means of increased power output. The variation in sea-level characteristics of available fuels is surprising. The influence of fuels on dependability is striking. It is hard to compare fuels from this point of view, without elaborate apparatus. It is impracticable to write into the specifications a test for anti-detonating qualities. Nevertheless, this somewhat intangible quality is one of the most important items in operation. No operator can afford to use any fuel but the best anti-knock fuel. There is much room for improvement here.

In any opinion, the compression-ignition type of engine has sufficient advantages from the point of view of reduction in fuel burned alone to warrant active development. Whether heavy fuel is burned by some new process, or is adapted to old forms of mechanism, remains to be seen.

There appears to me to be a possibility of great improvement through the use of new materials. A great deal is being learned about magnesium. The general employment of this material in aircraft appears to me to be not very far away. Magnesium has been found from the point of view of corrosion. It is very interesting, therefore, to learn that the production of magnesium has advanced to the point where this material is now being used in commercial chemical plants where no other material will withstand the corrosive agents encountered.

The present relation between the responsibility of the engineer and the propeller designer leaves a great deal to be desired. I believe, however, that progress lies in closer co-operation rather than through the attempt of any one of the branches to take over responsibility for all.

In general, it appears to me that there is room for, and hope for, engine design advancement in the near future, thus those successful in the year just closed.

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## Ascendancy of the Air-Cooled Radial— Diesels in Largest Sizes—

By LOUIS J. MEHRLE.

Chief Designer, Aviation Aircraft Engine Co.

AS REGARDS the trends of the aircraft engine industry in the year to come, we can have our predictions only on the developments which have come to light in the immediate past, and on such developments as our limited scope requires at the present.

Among the outside developments in the past year was the decrease in frontal resistance brought about by the introduction of high temperature cooling liquids. It has been assumed that several experimental projects are under way, which contemplate the use of high temperature cooling liquids both with the conventional liquid cooling arrangement of jackets, radiators and circulating pumps, and a second type in the field which will dissipate the heat from the cylinders through the cooling medium and directly from the fluid to the air by means of a forced

jet. It is the writer's opinion that although both types present numerous advantages over the water or low temperature liquid cooling, yet they present certain very definite disadvantages, which can only be eliminated by direct air cooling.

Without going into detail the main disadvantage seems to be increased weight per horsepower, increased cost and consequently reduced reliability and added possibilities of failure in the case of the types which still employ circulating pumps, possibilities of engine failure and subsequent forced landing due to excessive leakage either externally or, through cylinder head joints, internally. Proponents of liquid cooling and also of the in-line type of air-cooled engine will add the possibility of failure in the case of the types which still employ circulating pumps, possibilities of engine failure and subsequent forced landing due to excessive leakage either externally or, through cylinder head joints, internally. Proponents of liquid cooling and also of the in-line type of air-cooled engine will add the possibility of failure in the case of the types which still employ circulating pumps, possibilities of engine failure and subsequent forced landing due to excessive leakage either externally or, through cylinder head joints, internally. Proponents of liquid cooling and also of the in-line type of air-cooled engine will add the possibility of failure in the case of the types which still employ circulating pumps, possibilities of engine failure and subsequent forced landing due to excessive leakage either externally or, through cylinder head joints, internally.

In order to thoroughly understand the argument pro and con it is absolutely to get down to a few simple fundamentals. It is firstly, why it is necessary to cool an internal combustion engine? The answer is simply that the mean gas temperature within the cylinder and combustion chamber is higher than the critical temperature or the temperature at which known engineering materials lose the greater part of their strength.

In order, therefore, to reduce and maintain temperatures of cylinder and head below this critical temperature, it is necessary to bring about a temperature differential. In other words the outer surface of the metal must be exposed to a medium whose temperature is maintained far enough below the mean gas temperatures in contact with the inner surface so that the average temperature may fall below the critical temperature of the metal.

Thus, we have two factors determining the temperature of the cylinder and head metal. They are the mean gas temperature within the cylinder and the area of heating surface, and on the outside the area of cooling surface and the mean temperature of the fluid, whether it be liquid or gaseous, which is circulated over the cooling surface. Of course, there are other factors influencing the mean temperatures, such as the coefficients of heat conductivity within the fluids and the walls, as well as the coefficient of heat transfer from one to the other, but assuming that these three are relatively constant in both air- and liquid-cooled engines, we may eliminate them so that our discussion may not become too intricate. We may then say that the amount of heat transfer is determined by the product of the mean temperature of the engine under consideration. The heating surface is dependent on the bore stroke ratio and the combustion chamber design, but for purposes of comparison we may assume similar internal design.

With parallel amounts of heat available and heating surface, in order to maintain parallel mean temperatures we must provide for an equal amount of cooling and regardless of whether the heat passes directly from the metal to the air through cooling fins, or whether it goes from the metal to a fluid and thence to the air, either through a cooling radiator or a forced jet.

To the student of aerodynamics it becomes immediately obvious that the mass volume of air at the same temperatures and velocities may be passed over the cooling area in order to maintain reasonable cylinder

wall and heat temperatures regardless of the method of getting the heat from the heating surface to the air. This means that in two engines with the same horsepower the one with radial air-cooled, the other with in-line cylinders, either air- or liquid-cooled, the same relative amount of air must be circulated in order to provide the adequate cooling. In any opinion this gives the advantage of reduced drag to another type, but fairly and squarely divides the actual drag chargeable to cooling.

It is the personal opinion of the writer that the popularity of the radial versus the in-line V or X will be determined by considerations other than these mere technicalities.

There is at the present time a rather definitely cooled maximum due to which air cooled internal combustion cylinders may be built. This is probably due to the fact that the best available is in direct proportion to the volume and the volume increases as the cube of the mean dimensions whereas the area of the surface enclosing that volume increases only as the square of the mean dimensions. This prevents overloading problems which undoubtedly will be overcome eventually but are definitely limiting cylinder sizes at the present time.

It is also to be noted that demands for higher horsepower are rapidly increasing. This, due to the limitations of cylinders size, indicates the necessity of a greater number of cylinders per unit. It has been proven that size is the maximum number of cylinders practical to cool reliably by air cooling, so as to keep the mean gas radial engine capacities, having reached the maximum of nine cylinders, direct stream only two alternatives. The first is to develop larger cylinders and the second, to resort to the multi-bank radial. There are several instances of successful double bank radial turbines, but as far as we know there are no successful radial engines in existence today of more than two banks. This limitation may force engine designers to resort to inline multi-bank engine, however, we feel it highly improbable that engines of this size will appear in the immediate future.

Given the present outlook it would seem for some practical to devote research on designs above 800 hp to the Diesel type. It would seem from our limited experience that the Diesel due to the difficulty of metering sufficiently small quantities of fuel as the smaller engine and relatively greater in successful development to the larger types.

The most important development which we anticipate in the near future is the introduction of other new propeller designs and higher shaft speeds as a conventional propeller design, which will allow engine speeds to increase in the advantages of higher shaft speeds in order to reduce the unit engine weight. Although it is hardly consistent to speak of higher speeds and at the same time talk about greater reliability and increased life between overhauls, yet it is considered possible that by careful design and selection of materials having compensating parts may be developed which will produce about the same unit bearing loadings. Considering the general picture this should mean lower weight power plants for the same power output or greater power output for the same weight.

One further consideration in overall weight would mean less time at full throttle for the same ship performance, and in the second case a greater power per unit weight would mean either less time in covering a fixed distance or lower cruising time to cover the

same distance in the same time. All of which should mean increased life.

Regarding the possibility of increasing compression ratios, our attention is at the present time focused on two interesting solution systems: the down draft, and the more recent dual manifold which separates the intake strokes from the exhaust, recently introduced by Karl Kiesel of the Döhrle Laboratories. Both systems tend to lower induction temperatures, which in general indicates lower mean gas temperatures within the combustion space and should allow higher compression ratios without detonation.

It is the writer's opinion, however, that aside from supercharging, definitely higher compression ratios together with higher brake mean effective pressures and a general improvement in overall performance may be realized when fuel specifications become more definitely standardized and fuel of the standard specification becomes more generally available.

With regard to the possibilities of the use of the recently developed light alloys, particularly magnesium alloys, we believe it is very reasonable to expect a large number of manufacturers to standardize on the use of magnesium alloys for crankshafts, connecting bearings and all such engine parts which are directly subjected to extremely high heat loads. It is our understanding that magnesium has a higher coefficient of expansion and a decidedly lower coefficient on thermal conductivity, which would preclude its use in cylinder heads or piston cooling fins. It is our opinion that the time when the first heat treating will be radically less.

It is our opinion that the copper aluminum alloys known as Bimetal J or the 122 specification of the Aluminum Company show definite advantages in the way of better hardness and heat conductivity and will, therefore, be much more probably used in cylinder head and piston. Further developments will find increasing use in crankshafts and possibly in piston work, but it is somewhat doubtful what processes will be arrived at in the near future which will permit of forging the fins of a cylinder head although this is not at all to be considered an impossible task.

As regards engine mounting, it is hoped that in the not too distant future the engine manufacturer may be able to design, build and deliver a power unit to the airplane manufacturer, thus not to count on the engine designer and construction engineer, complete engine cooling, at least as far back as the normal engine mounting, and the engine mount as far back as the point at which it is normally attached to the four lugs of the fuselage. The construction of an exhaust pipe should be considered by the ship designer as a separate plant. The exhaust system should be left to the ship designer and the construction of the cooling between the normal engine mounting face and the point at which the engine mount joins the fuselage proper would normally be determined by the mounting contour of the ship, and also should be left to the ship designer, but the controls including spark, throttle, automatic doors, thermometer, oil pressure, gear position switch, fuel lines, cylinder head, thermocouples, etc., could be very easily designed to be brought back to the rear bulk head of the complete unit. The unit should also include such oil tankage and plumbing as is necessary to have a unobstructed flow well. This would immediately place an added burden on the engine manufacturer, but due to standardization and subsequent increased production should decrease the overall cost of the ship and greatly assist the industry.

## Sea-Level Supercharging—Better Reliability—Unified Responsibility for Installation

By LIEUT. COMDR. J. M. SHOEMAKER, U.S.N.  
*Naval Air Station, York, Delaware*

THE COMPARATIVE race is back for airplane designers. A wide choice of engine types, each offering definite advantages for a specific type of airplane. The fundamental requirements of an engine engine will not diminish. Dependability is of paramount importance, but the other fundamental requirements must be met for each particular case.

Liquid cooled powerplants, offer less drag and lower specific fuel consumption than air-cooled powerplants, but are heavier and more expensive. In small airplanes, the weight and powerplant cost are a large percentage of those of the complete airplane, the advantage would seem to be with the air-cooled engine. In large planes, powerplant weight and cost are a smaller percentage of the total.

And it becomes a question of whether or not the minimum powerplant drag and lower fuel consumption of the liquid-cooled powerplant more than offset the higher weight and cost. In this connection the powerplant drag can have such an effect on thrust horsepower required, that a low-drag powerplant can be lower powered for the same airplane performance than can a high-drag powerplant.

In the air-cooled engine field, radial and inverted Vee engines are available. They offer marked improvement (in loading, installation) in forward thrust and air flow, some decrease in powerplant drag over radial engines. But in loading installations the radial engine permits a concentration of powerplant weight as close as possible to the airplane center of gravity, which is a distinct advantage. In radial installations the other advantages, but so does the inline and Vee advantage of forward vision, past the engine.

Aircraft engines can be expected to make some extensive use of supercharging as a means of improving engine and airplane performance. The availability of a good compressible fresh medium is the one thing that prevents extensive application of superchargers today. Superchargers for boosting of engine power at sea level will probably be more widely used as the means of increasing cylinder pressures is less conducive to detonation than is increase in compression ratio. Satisfactory reductions in the specific weight of four-cycle engines do not seem immediately probable. It is more likely that maintenance of power engine weights coupled with improvement in materials and refinement of design, will increase reliability and durability.

The Bureau of Aeronautics has been fostering a policy of gradual improvement in aviation gasoline. Gasoline in 1930 will be the superior to that of former years as regards anti-knock characteristics and will have consistently available as well as reasonable price.

The early use of compression-ignition engines should properly be the field of high-powered powerplant tests in ships and large airplanes, where long flights will permit full advantage to be taken of the improvement in

fuel economy, and where the power plant weight will be a small part of the total aircraft structure.

Aircraft engine accessories such as pumps, magneto, propellers and carburetors are constantly improving. In addition, new methods in the engine accessory field are being encouraged in the hope that there will be competition throughout the field. The past has seen the accessory field dominated by a handful of dependable products, with the result that gross have been cut off sight. This has been a source of constant irritation to engine manufacturers, especially in the field of small engines in which the high-priced accessories have raised engine costs markedly.

Aircraft engine manufacturers are falling in line with the idea that they should supply their engine complete with exhaust collectors and cooling, and the large engine companies go further than that in the employment of installation engineers who check engine installations from both a powerplant and an aerodynamic viewpoint. The "idiot's rule" in this direction is the rule of the engine and propeller designer for specific installations.

• • •

## Closer Government Regulation—Work Toward Standardized Installation Details

By GLENN D. ANGLE

*Produceville, Indiana Aircraft Engine Corp.*

THREE CONSIDERATIONS from engineers intimately connected with the design and development of aircraft engines are available. They offer marked improvement (in loading, installation) in forward thrust and air flow, some decrease in powerplant drag over radial engines. But in loading installations the radial engine permits a concentration of powerplant weight as close as possible to the airplane center of gravity, which is a distinct advantage. In radial installations the other advantages, but so does the inline and Vee advantage of forward vision, past the engine.

I believe that aircraft engines up to 300 or possibly 500 hp will continue to be predominantly of the air-cooled radial types. Although this type apparently exists at the present moment in the lighter power ranges, I would hesitate to predict that it will continue to exist for an indefinite period in view of other developments which show promise even though their justification has yet to be proved. I refer particularly to the liquid-cooled types either liquid or air-cooled Diesel types, and air-cooled engines with inline, Vee or other forms of cylinder arrangements.

It would be altogether an assumption in my part to express detailed opinion on the possibilities of any of the above at this moment. Liquid cooling conditions, other than water have been known for quite a time and the modern industry and enthusiasm so noticeable at present seems a bit overdone. There is no question that in certain types of engines these developments are justified, but it will be some time before the outdoor cooling will become generally adopted as there are many problems to be solved.

We have listened to numerous discussions in regard to the use of compression-ignition engines employing heavier and denser fuels, and to date, however, have been down to discontinue some of the possibilities as well as to prove that the variety of problems associated with these developments are by no means simple ones. Great credit will be due any engineer or organization developing and producing compressible fresh medium in this category within the next few years. It is believed that the fullest advantage of the possibilities in this type can be obtained only in large units designed generally for lighter-than-air installations, and there is little, if any, hope of this type replacing the conventional engine for the lighter-than-air type of craft requiring output only 500 hp.

The inline, Vee, and other arrangements of cylinders in air-cooled engines have so few points in their favor when compared to the radial forms that there is no necessity of launching a detailed discussion. There are many other engine great hopes in the possibilities of air-cooled in types requiring the general direction of the cooling air but to be diverted. Some engines of this class may be developed to a fairly successful degree, but it is not expected that the popularity of the radial type will be seriously threatened in any power range. I look forward to marked improvements in the design and construction of air-cooled radial engines during the next year or so. The manufacturers of this type who have satisfactorily passed the preliminary stages of development and manufacturing can now devote their energies to the more detailed refinements of their product. This will exert itself in greater reliability, longer life, lighter weight, and possibly lower cost.

I anticipate greater use of aluminum, and although coatings of this material are more expensive at the present time, there is promise of a sufficient reduction in cost to place them in the line with aluminum, which larger elements are created. There are numerous other improvements in design, as well as developments of similar character, which further research and test will also place in general use. Developments of this nature, rather than new and untried types, will give the aircraft engine industry the desired results.

The trend toward higher engine speeds will continue, and along with this will be the further development of propeller reduction gears. This is in keeping with efforts to reduce the overall diameter of the radial engine, which seems to be the only serious objection raised against it, whether real or imagined. However, increased output without a proportionate increase in overall diameter is possible in the two-row type, and we may expect further developments along this line.

In addition to refinements in the engine, designers must give serious attention to the problem of installation. There has always been a noticeable lack of cooperation between the engine and plane manufacturer in matters of this kind. No doubt in due course there will be standards adopted for mountings, cowlings and the like, but first the airplane designers must arrive at a more common understanding regarding their requirements. When standardization is found to be practicable without suffering unnecessary hardships on the airplane designer, the engine manufacturer can and should furnish, or in any event accept, all sorts of the installation which might affect the performance of the engine.

At the present time there are a few distinct types of construction in air-cooled radial engines, and the majority of recent designs are merely copies. It is easily

conceived that a copy is never as good as the original. There should be some means of protection against piracy of the sort which would allow quality across that unjust litigation in the courts. Perhaps an association of leading manufacturers would serve such a purpose.

The Department of Commerce tests for approved type certificates have protected the public in a large extent against unscrupulous engineers. The regulations governing these tests have been revised occasionally to meet contingencies and not all the type certificates have been issued on a like basis. Too much regulation, as everyone knows, is extremely dangerous, but it does seem necessary for this department to function more rapidly in certain respects for the protection of consumers from unscrupulous. It is desirous and convincing for manufacturers to advertise the output of their engines greater than their approved rating. This should not be permitted, and if the manufacturer is not satisfied with the rating given the engine, he should be compelled to submit to another test if he wishes to sell his product at a higher rating.

Furthermore, as soon as the time and facilities will permit, it should become a duty of the Department of Commerce to re-assess a test as any engine which one of its representatives has passed from regular consideration. Beyond the point of approval of the manufacturer is maintaining proper workmanship and material specifications should not be permitted. Suggestions relating to design changes seem to be entirely out of order unless the test shows a change is necessary, and then that is the province of the manufacturer before being granted an approval.

Authority to regulate should extend beyond the black tests because of installations that do not provide conditions for which the engine is rated. I refer particularly to tests conducted on in-line air-cooled engines which require special air flow conditions when installed in a ship. It should be necessary to obtain cylinder temperature readings on all cylinders in flight, and these should not exceed a certain amount under given conditions, nor vary more than some prescribed figure. The same treatment should apply to radial air-cooled engines as well, and the engine manufacturer should be forced to consider under which the engine is actually being used. It is believed that greater uniformity of regulation on the order of the above remarks will go far toward stabilizing the industry, and is really will benefit the manufacturers of aircraft engines rather than restrict their activities in any real or slight degree.

• • •

## Refinements and More Speed—Commercial Considerations Control Design

By CARLAND P. FREED, Jr.

*Chief Pilot and Chief Engineer,  
Albion Aircraft Co.*

THE CRANSTON process (patented 1933), in an airplane, will be in the line of refinement of existing models, with some "frank" showings. Coming up of detail will accept the greater number of manufacturers, with more stress on interior and exterior. Aerodynamic design will increase with cleaner structures, better lines and more attention to streamlining of details. The structural details will be largely similar to the present, with more of the light alloys being used.





























## SIDE SLIPS

By  
Robert R. Osborn

FIFTEEN years aviation has had to compete with the radio for popularity with the general public. In the newspapers and the stock market. At a time like this, when the public seems to have become a little less air-minded and a little more radio-minded, it is tough to have a perfectly good news item taken away from us by a mis-guided headline writer.

"This U. S. Army Says no brass" was the heading placed on the following article: "W. E. Savage, senior engineer from the Material Division, U. S. Air Corps, is at the Boeing Aircraft Company conducting static tests on the new Boeing Army fighting plane."

Discovered by Mr. K. B. of Seattle, Washington, in the Seattle Post-Intelligencer.

Mrs. L. M. of Garden City, New York, protests bitterly against the fact that many people have become so sentimental over aviation that it seems to have arrived at the "holy talk" stage. She quotes from an article in the New York Times discussing the recent when-meeting of the U. S. Army First Pursuit Group: "Flying much of the time in temperatures well below zero, facing day after day of storms and tailoring down frays, fuses, fuses, fuses and fuses, thousands of the vigorous little Coast Black porcupine planes made the tour and came through in good shape."

"How," asks Mrs. L. M., "can we keep the little birds and fuses of aviators this little birds warm?"

Everyone knows that airplane production conditions are in fairly bad shape just now, but that's no reason for the newspapers to get sarcastic and funny about a serious matter. H. E. W. reports the following clipping from the New York edition of the *Dayton Times*:

"New York, Jan. 10th.—The Ford Motor Co. was reported today will resume capacity operations of its airplane manufacturing division on February 1st."

## OUR HANGER FIVE DEPARTMENT

The following story by G.P.H. comes to us through S.T. of the Roosevelt Field Flying Service, who was just as much impressed by the story as we were. It seems that G.P.H. used to fly the Boeing out and chase that sort of the windy weather obtainable in Denver under the worst words anywhere else seem like grade spring repliers. The words were so bad in fact that they weren't able to devise a wind sock which would stand up until they finally made one up out of a three foot length of steel tire chain. When this wind sock was blown straight out it was still all right to fly, but, so far, they had always called off flying when the wind started snapping links off of the chain.

We seem to be getting more and more evidence from all sides to prove that coons are completely air-minded. It has been estimated that so far some time. Further advancement on the gastrointestinal delights that aviation holds for the humble coon come from E. W. R. of Cincinnati, Ohio.

"Your own stories in the January 11th issue do not do justice to the apparent fondness of the lady before for goodfies. Some years ago two Cincinnati prisoners were bringing a plane to Cincinnati from Memphis. Engine trouble caused a forced landing in an Indiana pasture. Thinking it might be dirty gasoline causing the engine trouble, they drained the tank into a bucket from their lot, and vomited the nearest air station for a fresh supply. In the meantime Benny sighted the bucket and captured the contents into one of his several stomachs. When they returned with fresh gas, they found a dead coon and a farmer's wife with a constable who demanded payment for the coon. The money was forthcoming, the constable took them to the officer who had permitted them to "phone Cincinnati for relief." C. E. Lee, manager of the coon, after considerable trouble, put the money on

the way by wire and they were released."

You remember the famous story about the mysterious accident that happened to a ship which had been designed and was being tested by the Army Engineering Division at McCook Field—the ship that was left standing on the field while the crew went out to lunch, and when they returned the landing gear had collapsed? Well, we think we have a possible solution for the mystery at last—a crew ate the bolts out of the fittings.

W. K. of Daytona Beach, Florida, quotes the following from an interview given to the reporters by a young lady who has just returned from looking over the aeronautical situation in England.

"The greatest difference between the American and British ships does not lie in the outward appearance but in the inward working of the engine. This is powerfully illustrated by the action of the different ships when out of control. The British ships fall head on in a 'Spinning Dive Dive' while the American planes—well, headover into the ditch 'Tail Spin'."

We don't take this as a reflection on American manufacturers, we'd say, it was just another example of the home way they do things in England—that engine having ships that would fall head on in a Spinning Nose Dive. Sigh, we sigh it.

My G. K. G. of Lakewood, Ohio, sends us a clipping from a contemporary aeronautical magazine which quotes one of its writers as having been "Chief Merchandiser of the American Cotton Ginning, Inc."

Readers will be surprised to make their own comments in this case. If the readers will only make a few comments for themselves, say one at first, but taking in at least one more comment each week, possibly AVIATION can do without this column altogether very soon.

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